

StarBooster™

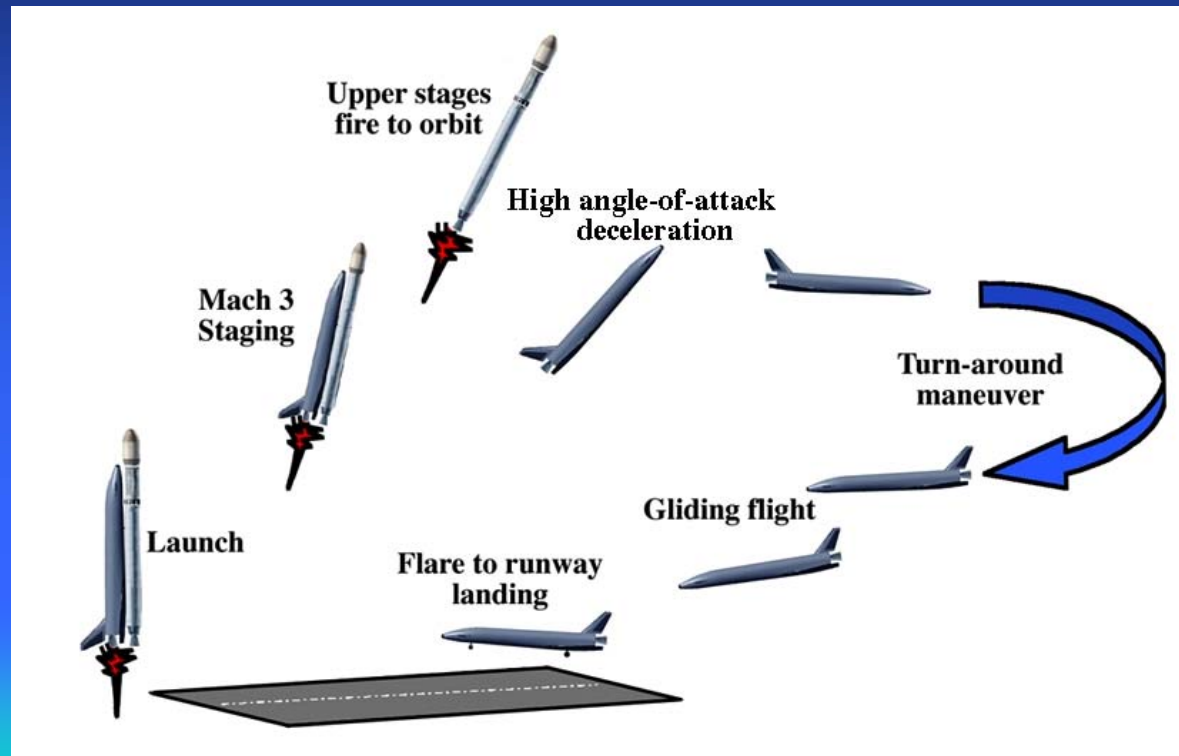
**The Optimum Reusable,
Rapid Response, Affordable, and Effective
U.S. Military Space Launch Capability**

Dirk Jameson, Lt.Gen. USAF (Ret)

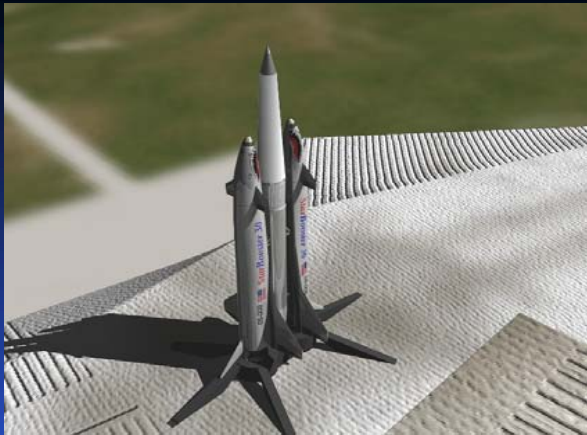
March 2006

What is *StarBooster*TM ?

- **Reusable First Stage - Return to launch site after low-Mach number ($M \sim 3$) staging**
 - Evolutionary approach to reusability – Learn by experience
 - Current jet-fighter technologies - operationally effective - not technology driven (the downfall of many other systems)



StarHawk Operational System



StarHawk on alert status



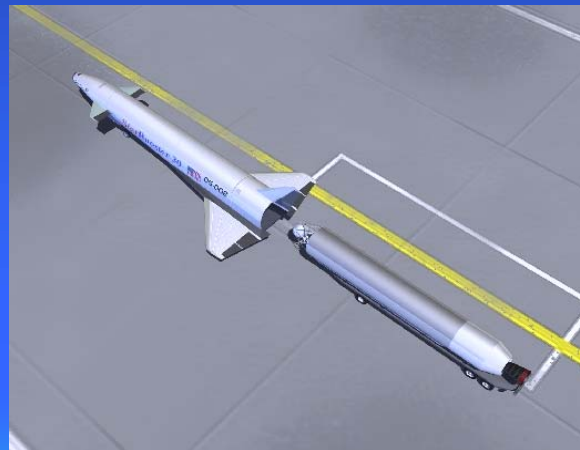
StarBooster separation - Mach 3



AFRL upper stage separation



StarBooster return



StarBooster servicing



AFRL upper stage recovery option

StarHawk Squadron at CONUS Base

One pad for routine operations,
provides war fighter with surge
capability

Two *StarHawks* on pad alert
Strike anywhere within one hour
Recover and Recycle in 12 hours

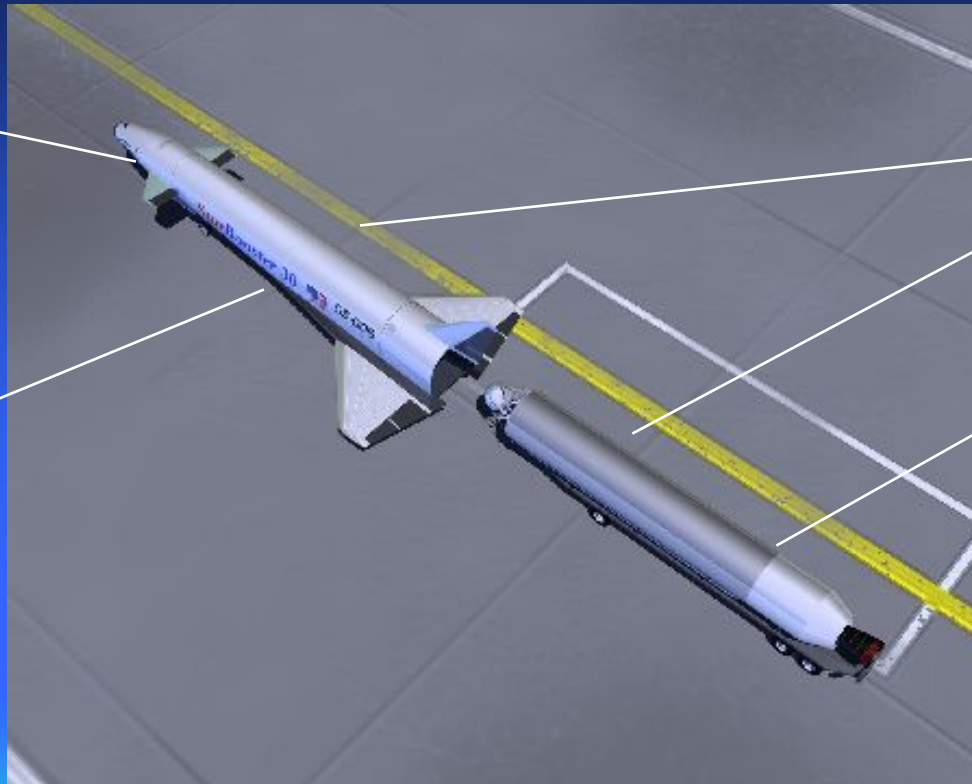


*StarBooster*TM Elements

- **Operable - Separable airframe and *Reusable Propulsion Module*TM - a key to easy access, fast maintenance, quick turnaround, rapid response**

Panel access
to modularized
subsystems

Low-maintenance
aluminum airframe
Low landing speeds



Mate any available
airframe and
propulsion module
for rapid turnaround

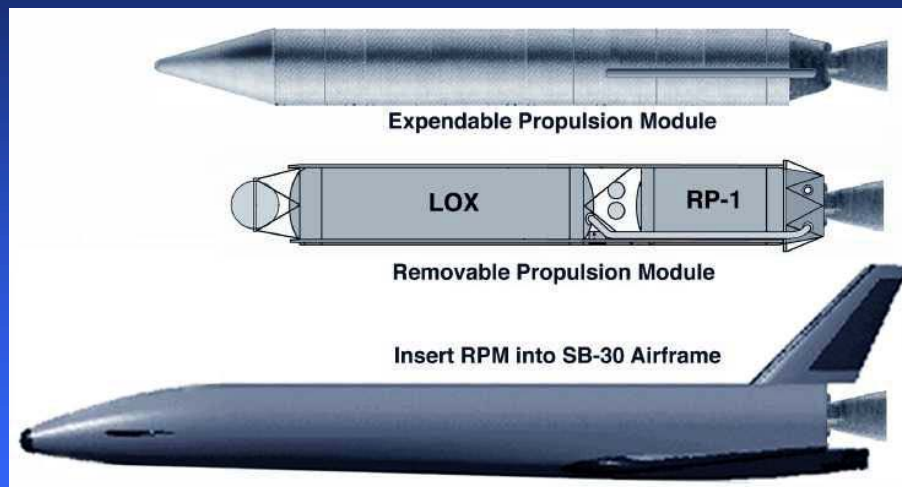
Separate propulsion
inspection,
maintenance

Kerosene fuel &
liquid oxygen loaded
on the pad

StarHawk Modular Approach

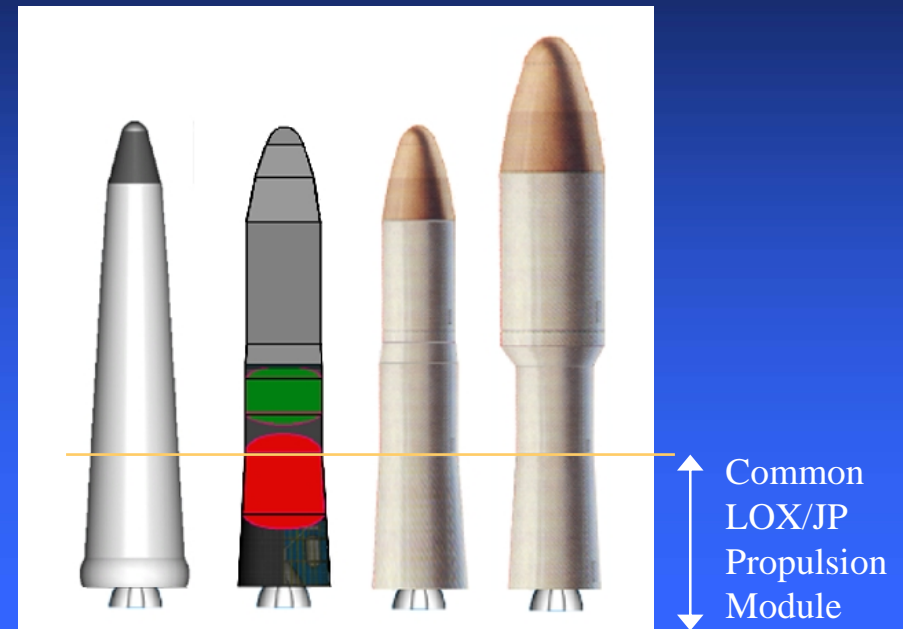
- **Modular**

- Utilize common elements to minimize development, production and operations costs and allow flexibility in missions & operations



Common design & production

- *RPM* for booster
- *EPM* for core stage; drop tanks



Upper stage modularity

Reusable & Expendable Mission Options
(Reusable AFRL-VS based design)

Application to Growth Systems

- Concept scalable to meet multiple mission requirements
- Common booster & core stage designs



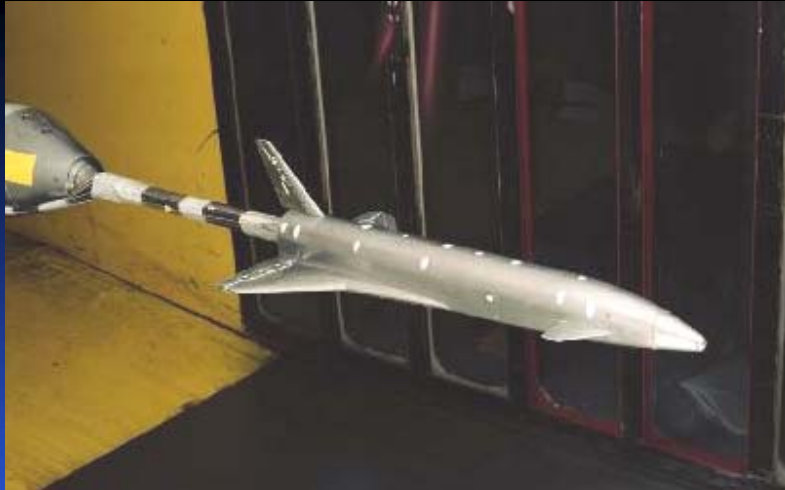
Demo

SLV
1,000 lbm

StarHawk I
to 6,000 lbm

StarHawk II
to 25,000 lbm

Early Testing to Date

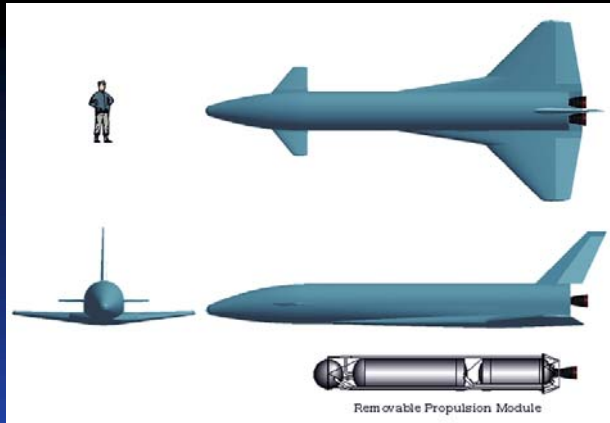


Langley Research Center Langley Glideback Booster (LGBB) Wind Tunnel Tests

- Supersonic tests up to Mach 4.6
 - Determine control effectiveness
 - Support aero database development
 - Support glideback analysis & stability/control design
 - Support staging model



Demonstrator Development Program



Selected smaller scale engineering & risk mitigation activities
– Facility Verification Vehicle Projects

Ground Test Propulsion Module



+



Subsonic Landing Test Vehicle

3 Supersonic
StarBooster 5 Test
Vehicles with RPMs
Flight Test Program to
Mach 3+



Integrated *StarHawk*
Configuration Tests -
Includes subsonic staging
dynamics tests
and dual *StarBooster*
glide backs & landing



Comparisons

- *StarBooster* not technology driven (e.g. unrealized expectations of high tech *Space Shuttle* and *X-33*, *NASP* projects)
- Expendable systems more costly as flight rates increase (and costly still at low flight rates - e.g. *Titan IV* and *EELV*)
- *StarBooster* stages at low-Mach numbers:
Hypersonic airbreathers with hypersonic staging raise costly development & technology issues and complicates operations
- *StarHawk* system based on *StarBooster* modular approach provides optimum mix of reusability and expendability to minimize development and required technologies while minimizing recurring costs of operations
- *StarHawk* system modularity also provides tailoring of elements used for specific mission needs - not an option in many other approaches